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CORRELATION OF HELICOPTER NOISE LEVELS WITH PHYSICAL AND PERFOR--ETC(U)  
SEP 80 J S NEWMAN  
FAA/EE-80-42

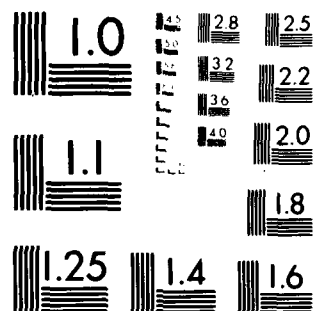
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U.S. Department  
of Transportation  
**Federal Aviation  
Administration**

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## Correlations of Helicopter Noise Levels with Physical and Performance Characteristics

Office of Environment  
And Energy  
Washington, D.C. 20591

### A Preliminary Report

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By J. Steven Newman

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<p>This report investigates the correlation between physical and performance characteristics of helicopters and the noise levels which they generate in various operational modes. The analysis is generally empirical although several theoretical functions described in the literature have been examined. The EPNL is the acoustical metric employed in this study. One, two, and three-step multiple regression analyses are conducted for takeoff, approach, and level flyover operations. Plots are provided for the three best single variable regression models for each mode of flight.</p>			
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## 1.0 INTRODUCTION

This report investigates the correlation between physical and performance characteristics of helicopters and the noise levels which the helicopters generate in various operational modes. The analysis is generally empirical although several theoretical functions described in the technical literature have been examined. The Effective Perceived Noise Level (EPNL) is the acoustical metric employed in this study. It is anticipated that subsequent analyses will examine trends for other metrics, in particular the Noise Exposure Level (single event, integrated A-Weighted Sound Level). This report has been limited to presenting statistical analyses. Parameters are tested for correlation with EPNL, a single event cumulative energy noise metric. The units of most analysis parameters are not energy. However, a limited number of variables dimensionally consistent with power, intensity or energy are tested.

### 1.1 Input Data Files

This study utilizes a data file assembled for analyses conducted in FAA-AEE-79-3, "Noise Levels and Flight Profiles of Eight Helicopters Using Proposed International Certification Procedures (Newman, J. S., Rickley, E. J.). This data file (Table 1.1) has been expanded to include a variety of physical parameters which may be expected to influence resulting noise levels. The legend for Table 1 is presented below:

NOTE:  $V_H$  is the speed at maximum continuous power.  
 $V_{NE}$  is the never exceed speed.

LEGEND FOR CROSS CORRELATION MATRIX

TYPE	-	Helicopter designation
EPNL	-	Effective Perceived Noise Level, (level flyover) expressed in decibels
WEIGHT	-	Test weight, lbs.
AREA	-	Total main rotor blade area (square feet)
MACH	-	Mach number of advancing blade; sum .9 (lesser of $V_H$ or $V_{NE}$ ) level flyover forward speed and rotational tip speed
SHP	-	Maximum engine shaft horsepower
M-DISC	-	Main disc area, square feet
BRC	-	Best rate of climb, feet/minute
M-FREQ	-	Main rotor blade frequency. Using main rotor rpm (Jane's) and number of blades (Jane's) units in Blade Passages/Sec.
T-BLADE	-	Total tail rotor blade area (square feet)
EPNLA	-	Effective Perceived Noise Level (approach), decibels
LOGW	-	Common logarithm of weight
LOGA	-	Common logarithm of area
LOGS	-	Common logarithm of shaft horsepower
LOGMD	-	Common logarithm of main disc area
DISPLO	-	Main disc loading, lbs/square feet
LOGTB	-	Common logarithm of total tail blade area
MACH6	-	Mach number to sixth power
TSPEED	-	Rotational tip speed of tail rotor (feet/second)
MSPEED	-	Main rotor rotational tip speed (feet/second)
T-MACH	-	Tail rotor, rotational tip mach number
F1	-	$\text{Log}_{10} ((T-MACH \times \text{Weight})^2 / T-Blade)$

- F2 -  $\text{Log}_{10} ((\text{MACH} \times \text{Weight})^2 / \text{AREA})$
- F3 -  $\text{Log}_{10} (\text{SHP} \times \text{M-DISC} / \text{MACH})$
- F4 -  $\text{Log}_{10} (\text{MACH}^6 \times \text{T-BLADE})$
- F5 -  $\text{Log}_{10} (\text{T-MACH}^6 \times \text{AREA})$

## 2.0 CROSS CORRELATION OF ANALYSIS PARAMETERS

This section examines and discusses the interdependence between the parameters used in the regression analyses.

### 2.1 Cross Correlation Matrix

The matrix displayed as Table 2.1 (four pages) provides insight into the interrelationships between test variables and the relationship of each variable to EPNL for takeoff, approach and level flyover. Each entry in the matrix includes the correlation coefficient "R," the probability that the observed correlation is due merely to chance, and the number of observations. Many of the variables correlate equally well with the acoustical measures for all operational modes. Therefore, it is possible to develop a large number of single variable regression models which predict noise with similar accuracy.

The "good predictor" family of parameters includes:

- Weight: Helicopter Weight
- Area: Main Rotor Area
- SHP: Shaft Horsepower
- MDISC: Main Rotor Disc Area
- LOGW: Log Weight
- LOGA: Log Area

# 1.1 INPUT DATA FILES

## HELICOPTER NOISE REGRESSION ANALYSIS

OBS	TYPE	EPNL	WEIGHT	AREA	MSPEED	SHP	M DISC	BRC	M FREQ	T BLADE	EPNLA	EPNLT	TSPEED
1	SA330J	91.4	15,532	188	682	3,150	1,925	1,400	17.6	15.0	95.6	95.4	669
2	B0105	88.4	5,070	116	724	840	804	1,772	22.5	3.7	91.7	89.1	730
3	B206L	85.8	4,000	37	763	840	1,075	1,550	13.1	2.2	90.3	85.9	691
4	S61	92.6	22,050	232	659	2,800	3,019	2,200	16.9	12.1	94.0	95.9	630
5	S65	97.1	37,000	475	700	7,925	4,070	2,180	18.5	19.2	99.9	95.7	663
6	B212	94.6	10,500	82	817	1,800	1,809	1,420	10.8	8.1	95.7	91.7	740
7	H500C	85.8	2,550	29	667	400	547	1,700	32.3	1.7	87.7	85.1	692
8	SA341G	86.1	3,970	51	681	590	931	1,770	18.9	1.9	89.5	89.5	696
9	SA350	87.2	4,180	47	671	641	962	1,575	18.3	3.7	91.2	89.2	653
10	SA321F	92.0	25,300	298	688	4,650	3,058	1,312	21.2	29.5	98.6	98.4	351
11	A109	90.4	5,390	80	727	840	1,023	1,620	25.7	4.4	93.0	-	728
12	M16A	103.4	88,440	942	721	11,000	10,386	1,476	10.0	64.9	107.4	-	733
13	M12	89.5	7,755	94	623	900	1,772	1,320	12.4	6.5	96.1	-	631
14	M18	97.3	25,212	298	702	3,000	3,848	1,102	16.0	16.0	99.6	-	732
15	WG-13	97.7	9,350	109	699	1,500	1,385	1,200	21.2	8.6	96.9	91.6	717
16	B47G	90.3	2,728	33	720	220	1,075	800	12.4	2.4	89.6	-	659
17	H300C	80.6	1,804	23	665	190	566	750	23.7	1.7	-	-	689
18	S64	96.7	42,812	518	700	9,000	4,070	1,330	18.6	20.2	98.6	-	702
19	SA365C	89.4	7,480	83	689	1,300	1,116	1,968	23.3	9.8	94.0	-	725

1.1 INPUT DATA FILES  
(Continued)

HELICOPTER NOISE REGRESSION ANALYSIS

OBS	TYPE	LOGW	LOGA	LOGTB	LOGS	LOGMD	DISLO	MACH	MACH6
1	SA330J	4.19	2.27	1.18	3.50	3.28	8.07	0.80	0.26
2	B0105	3.71	2.06	0.57	2.92	2.91	6.31	0.83	0.32
3	B206L	3.60	1.57	0.34	2.92	3.03	3.72	0.86	0.41
4	S61	4.34	2.37	1.08	3.45	3.48	7.30	0.79	0.24
5	S65	4.57	2.68	1.28	3.90	3.61	9.09	0.82	0.31
6	B212	4.02	1.91	0.91	3.26	3.26	5.80	0.87	0.45
7	H500C	3.41	1.46	0.23	2.60	2.74	4.67	0.78	0.22
8	SA341G	3.60	1.71	0.28	2.77	2.97	4.26	0.80	0.27
9	SA350	3.62	1.67	0.57	2.81	2.98	4.35	0.77	0.21
10	SA321F	4.40	2.47	1.47	3.67	3.49	8.27	0.79	0.25
11	A109	3.73	1.90	0.64	2.92	3.01	5.27	0.85	0.37
12	M16A	4.95	2.97	1.81	4.04	4.02	8.52	0.83	0.32
13	M12	3.89	1.97	0.81	2.95	3.25	4.38	0.67	0.09
14	M18	4.40	2.47	1.20	3.48	3.59	6.56	0.79	0.25
15	WG-13	3.97	2.04	0.93	3.18	3.14	6.75	0.82	0.31
16	B47G	3.44	1.52	0.38	2.34	3.03	2.54	0.77	0.21
17	H300C	3.26	1.36	0.23	2.28	2.75	3.19	0.72	0.14
18	S64	4.63	2.71	1.31	3.95	3.61	10.52	0.78	0.22
19	SA365C	3.87	1.92	0.99	3.11	3.05	6.70	0.80	0.27

1.1 INPUT DATA FILES  
(Concluded)

HELICOPTER NOISE REGRESSION ANALYSIS

OBS	TYPE	F1	F2	F3	F4	F5
1	SA335	6.76189	5.91467	6.87955	1.69328	0.94073
2	B0105	6.47313	5.18251	5.91106	1.57532	0.95841
3	B226L	6.44533	5.50604	6.02063	1.17856	0.31908
4	S61	7.10738	6.11200	7.03169	1.73750	0.87555
5	S65	7.40080	6.29196	7.59247	2.17344	1.31979
6	B212	6.77703	6.01214	6.57093	1.56453	0.84322
7	H500C	6.16752	5.13039	5.45019	0.80152	0.21705
8	SA3416	6.50872	5.30027	5.83467	1.13834	0.47724
9	SA350	6.20865	5.34392	5.90320	0.99310	0.27559
10	SA321F	6.33170	6.13156	7.25312	1.87283	-0.53993
11	A109	6.44866	5.41562	6.00639	1.46970	0.78989
12	M16A	7.71593	6.75620	8.13936	2.48492	1.87869
13	M12	6.47098	5.45619	6.37763	0.92358	0.48732
14	M18	7.23279	6.12854	7.16259	1.87283	1.37530
15	HG-13	6.62283	5.73361	6.40284	1.52566	0.88456
16	B47G	6.03392	5.12583	5.48750	0.83649	0.14584
17	H300C	5.86313	4.86593	5.17393	0.50728	0.10506
18	S64	7.55513	6.32951	7.67348	2.05646	1.50637
19	SA365C	6.38192	5.63898	6.25648	1.34985	0.79512

TABLE 2.1

CORRELATION COEFFICIENTS / PROB. $\geq 1\%$ UNDER $H_0: RHO=0$ / NUMBER OF OBSERVATIONS													
	EPNL	WEIGHT	AREA	MSPEED	SHP	M_DISC	BRC	M_FREQ	T_BLADE	EPNLA	EPNLT	TSPEED	LOGM
EPNL	1.00000 0.0000 0.0000	0.79053 0.0001 0.0001	0.79117 0.0001 0.0001	0.25695 0.2883 0.2883	0.77085 0.0001 0.0001	0.79269 0.0001 0.0001	0.08050 0.7432 0.7432	-0.43628 0.0618 0.0618	0.74421 0.0003 0.0003	0.91138 0.0001 0.0001	0.67551 0.0225 0.0225	0.09015 0.7136 0.7136	0.87708 0.0001 0.0001
WEIGHT	0.79053 0.0001 0.0001	1.00000 0.0000 0.0000	0.99371 0.0001 0.0001	0.07690 0.7544 0.7544	0.94703 0.0001 0.0001	0.98737 0.0001 0.0001	0.10075 0.6815 0.6815	-0.39223 0.0967 0.0967	0.95648 0.0001 0.0001	0.87924 0.0001 0.0001	0.84780 0.0010 0.0010	-0.01080 0.9650 0.9650	0.87139 0.0001 0.0001
AREA	0.79117 0.0001 0.0001	0.99371 0.0001 0.0001	1.00000 0.0000 0.0000	0.05399 0.8262 0.8262	0.96552 0.0001 0.0001	0.97105 0.0001 0.0001	0.12136 0.6206 0.6206	-0.34309 0.1504 0.1504	0.93830 0.0001 0.0001	0.88024 0.0001 0.0001	0.80291 0.0029 0.0029	-0.02204 0.9286 0.9286	0.88403 0.0001 0.0001
MSPEED	0.25695 0.2883 0.2883	0.07690 0.7544 0.7544	0.94703 0.0001 0.0001	1.00000 0.8262 0.8262	0.96552 0.0001 0.0001	0.97105 0.0001 0.0001	-0.05065 0.8369 0.8369	-0.28462 0.2376 0.2376	0.06807 0.7819 0.7819	0.05264 0.8357 0.8357	-0.19970 0.5560 0.5560	0.31234 0.1930 0.1930	0.06854 0.7804 0.7804
SHP	0.77085 0.0001 0.0001	0.0001 0.0001 0.0001	0.0001 0.0001 0.0001	0.0000 0.7469 0.7469	0.77085 0.0001 0.0001	0.79269 0.0001 0.0001	0.08050 0.7432 0.7432	-0.43628 0.0618 0.0618	0.74421 0.0003 0.0003	0.91138 0.0001 0.0001	0.67551 0.0225 0.0225	0.09015 0.7136 0.7136	0.87708 0.0001 0.0001
M_DISC	0.79269 0.0001 0.0001	0.0001 0.0001 0.0001	0.0001 0.0001 0.0001	0.0000 0.7469 0.7469	0.77085 0.0001 0.0001	0.79269 0.0001 0.0001	0.08050 0.7432 0.7432	-0.43628 0.0618 0.0618	0.74421 0.0003 0.0003	0.91138 0.0001 0.0001	0.67551 0.0225 0.0225	0.09015 0.7136 0.7136	0.87708 0.0001 0.0001
BRC	0.08050 0.7432 0.7432	0.10075 0.6815 0.6815	0.94703 0.0001 0.0001	0.07690 0.7544 0.7544	0.94703 0.0001 0.0001	0.98737 0.0001 0.0001	0.10075 0.6815 0.6815	-0.39223 0.0967 0.0967	0.95648 0.0001 0.0001	0.87924 0.0001 0.0001	0.84780 0.0010 0.0010	-0.01080 0.9650 0.9650	0.87139 0.0001 0.0001
M_FREQ	-0.43628 0.0618 0.0618	-0.39223 0.0967 0.0967	-0.34309 0.1504 0.1504	-0.28462 0.2376 0.2376	-0.31874 0.1835 0.1835	-0.43733 0.95913 0.95913	-0.06522 0.7908 0.7908	-0.43733 0.95913 0.95913	-0.06522 0.7908 0.7908	-0.43733 0.95913 0.95913	-0.06522 0.7908 0.7908	-0.43733 0.95913 0.95913	-0.06522 0.7908 0.7908
T_BLADE	0.74421 0.0003 0.0003	0.91138 0.0001 0.0001	0.87924 0.0011 0.0011	0.88024 0.0001 0.0001	0.84641 0.0001 0.0001	0.83209 0.0001 0.0001	-0.09928 0.6951 0.6951	-0.41936 0.88060 0.88060	1.00000 0.0000 0.0000	0.88060 0.0001 0.0001	0.90556 0.0001 0.0001	-0.16209 0.5074 0.5074	0.82817 0.0001 0.0001
EPNLA	0.91138 0.0001 0.0001	0.87924 0.0011 0.0011	0.88024 0.0001 0.0001	0.84641 0.0001 0.0001	0.83209 0.0001 0.0001	-0.09928 0.6951 0.6951	-0.41936 0.88060 0.88060	1.00000 0.0000 0.0000	0.88060 0.0001 0.0001	0.90556 0.0001 0.0001	-0.16209 0.5074 0.5074	0.82817 0.0001 0.0001	0.82817 0.0001 0.0001
EPNLT	0.67551 0.0225 0.0225	0.84780 0.0010 0.0010	0.80291 0.0029 0.0029	-0.19970 0.5560 0.5560	0.76887 0.0057 0.0057	0.85679 0.0006 0.0006	0.08042 0.8142 0.8142	-0.08042 0.8142 0.8142	0.90556 0.0001 0.0001	0.85292 1.00000 1.00000	0.0008 0.0000 0.0000	-0.07047 0.60426 0.60426	0.93104 0.0001 0.0001
TSPEED	0.09015 0.7136 0.7136	-0.01080 0.9650 0.9650	-0.02204 0.9286 0.9286	0.31234 0.1930 0.1930	-0.08031 0.7438 0.7438	-0.00042 0.9986 0.9986	0.06069 0.8051 0.8051	-0.00522 0.9831 0.9831	-0.16209 0.5074 0.5074	-0.07047 0.60426 0.60426	1.00000 0.0000 0.0000	1.00000 0.0000 0.0000	-0.14994 0.5401 0.5401
LOGM	0.87708 0.0001 0.0001	0.87139 0.0001 0.0001	0.88408 0.0001 0.0001	0.06854 0.7804 0.7804	0.89840 0.0001 0.0001	0.84931 0.0001 0.0001	0.23246 0.2382 0.2382	-0.39572 0.0935 0.0935	0.82317 0.0001 0.0001	0.93104 0.0001 0.0001	0.94381 0.0001 0.0001	-0.14994 0.5401 0.5401	1.00000 0.0000 0.0000

TABLE 2.1 (Continued)

	CORRELATION COEFFICIENTS / PROB > IRI UNDER H0:RHO=0 / NUMBER OF OBSERVATIONS												F5
	LOGA	LOGTB	LOGS	LOGMD	DISLO	MACH	TMACH	MACH6	F1	F2	F	F4	F5
EPNL	0.85276 0.0001 19	0.85340 0.0001 19	0.82731 0.0001 19	0.87489 0.0001 19	0.72512 0.0004 19	0.33729 0.1579 19	0.09015 0.7136 19	0.30776 0.1999 19	0.84583 0.0001 19	0.88044 0.0001 19	0.86046 0.0001 19	0.86563 0.0001 19	0.77495 0.0001 19
WEIGHT	0.85554 0.0001 19	0.82667 0.0001 19	0.81266 0.0001 19	0.89622 0.0001 19	0.70275 0.0008 19	0.16211 0.5073 19	-0.01080 0.9650 19	0.12237 0.6177 19	0.83365 0.0001 19	0.84928 0.0001 19	0.86558 0.0001 19	0.81501 0.0001 19	0.67820 0.0014 19
AREA	0.89427 0.0001 19	0.83270 0.0001 19	0.83315 0.0001 19	0.89637 0.0001 19	0.74307 0.0003 19	0.15438 0.5280 19	-0.02204 0.9286 19	0.10794 0.6600 19	0.84829 0.0001 19	0.84584 0.0001 19	0.87861 0.0001 19	0.83862 0.0001 19	0.67488 0.0015 19
MSPEED	0.01717 0.9444 19	0.03924 0.8733 19	0.11961 0.6257 19	0.08541 0.7231 19	0.01727 0.9440 19	0.82387 0.0001 19	0.31234 0.1930 19	0.87711 0.0001 19	0.15912 0.5153 19	0.19667 0.4197 19	0.08299 0.7355 19	0.26621 0.2706 19	0.23625 0.3302 19
SHP	0.88390 0.0001 19	0.83566 0.0001 19	0.88327 0.0001 19	0.88437 0.0001 19	0.82687 0.0001 19	0.17251 0.4800 19	-0.08031 0.7438 19	0.12451 0.6115 19	0.85673 0.0001 19	0.86918 0.0001 19	0.90433 0.0001 19	0.24945 0.0001 19	0.63452 0.0035 19
M_DISC	0.82507 0.0001 19	0.81062 0.0001 19	0.77031 0.0001 19	0.90113 0.0001 19	0.61882 0.0047 19	0.14875 0.5433 19	-0.00042 0.9986 19	0.12152 0.6202 19	0.81099 0.0001 19	0.83452 0.0001 19	0.84173 0.0001 19	0.78227 0.0001 19	0.67330 0.0016 19
BRC	0.24999 0.3020 19	0.14655 0.5494 19	0.32369 0.1764 19	0.11520 0.6386 19	0.34516 0.1478 19	0.34802 0.1443 19	0.06069 0.8051 19	0.31324 0.1916 19	0.31043 0.1958 19	0.23907 0.3243 19	0.23710 0.3284 19	0.33174 0.1653 19	0.18344 0.4522 19
M_FREQ	-0.28834 0.2312 19	-0.36277 0.1269 19	-0.31353 0.1912 19	-0.56503 0.0117 19	-0.02767 0.9105 19	-0.03070 0.9007 19	-0.00522 0.9831 19	-0.11339 0.6439 19	-0.40076 0.0891 19	-0.47037 0.0421 19	-0.42490 0.0698 19	-0.26119 0.2301 19	-0.32912 0.1689 19
T_BLADE	0.80352 0.0001 19	0.84837 0.0001 19	0.76764 0.0001 19	0.85189 0.0001 19	0.65043 0.0026 19	0.16482 0.5001 19	-0.16209 0.5074 19	0.12653 0.6057 19	0.69194 0.0010 19	0.81245 0.0001 19	0.81943 0.0001 19	0.77375 0.0001 19	0.56792 0.0112 19
EPNLA	0.91026 0.0001 18	0.94305 0.0001 18	0.88246 0.0001 18	0.92922 0.0001 18	0.73837 0.0005 18	0.06959 0.7838 18	-0.07047 0.7811 18	0.07347 0.7720 18	0.80155 0.0001 18	0.90649 0.0001 18	0.92477 0.0001 18	0.25568 0.0001 18	0.69888 0.0013 18
EPNLT	0.92220 0.0001 11	0.95400 0.0001 11	0.90912 0.0001 11	0.90917 0.0001 11	0.88556 0.0003 11	-0.13522 0.6918 11	-0.60426 0.0490 11	-0.16689 0.6238 11	0.57894 0.0620 11	0.86743 0.0005 11	0.92252 0.0001 11	0.26066 0.0007 11	0.19134 0.5730 11
TSPEED	-0.14272 0.5600 19	-0.23415 0.3346 19	-0.15392 0.5293 19	-0.15240 0.5334 19	-0.14511 0.5534 19	0.26467 0.2735 19	1.00000 0.0000 19	0.29416 0.2215 19	0.20866 0.3913 19	-0.12001 0.6246 19	-0.16626 0.4963 19	-0.04759 0.8466 19	0.58925 0.0079 19
LOGW	0.98145 0.0001 19	0.96313 0.0001 19	0.97745 0.0001 19	0.96936 0.0001 19	0.88872 0.0001 19	0.21242 0.3826 19	-0.14994 0.5401 19	0.17339 0.4778 19	0.90740 0.0001 19	0.97752 0.0001 19	0.99640 0.0001 19	0.24254 0.0001 19	0.69491 0.0012 19



TABLE 2.1 (Continued)

	CORRELATION COEFFICIENTS / PROB : IRI										OBSERVATIONS				LOGM
	EPNL	WEIGHT	AREA	MSPEED	SHP	M_DISC	BRC	M_FREQ	T_BLADE	EPNLA	EPNLT	TSPEED			
LOGA	0.85276 0.0001	0.85554 0.0001	0.88427 0.0001	0.01717 0.2444	0.88990 0.0001	0.82507 0.0001	0.24999 0.3020	-0.28834 0.2312	0.80852 0.0001	0.91026 0.0001	0.92220 0.0001	-0.14272 0.5600	0.98145 0.0001		
LOGTB	0.85340 0.0001	0.82667 0.0001	0.83270 0.0001	0.03924 0.8733	0.83566 0.0001	0.81062 0.0001	0.14655 0.5494	-0.36277 0.1269	0.84837 0.0001	0.94305 0.0001	0.95400 0.0001	-0.23415 0.3346	0.96313 0.0001		
LOGS	0.82731 0.0001	0.81266 0.0001	0.83315 0.0001	0.11961 0.6257	0.88327 0.0001	0.77031 0.0001	0.32369 0.11764	-0.31353 0.1912	0.76764 0.0001	0.98246 0.0001	0.90912 0.0001	-0.15392 0.5293	0.97745 0.0001		
LOGMD	0.87489 0.0001	0.89622 0.0001	0.89637 0.0001	0.08541 0.7281	0.88437 0.0001	0.90113 0.0001	0.11520 0.6386	-0.56503 0.0117	0.85189 0.0001	0.92922 0.0001	0.90917 0.0001	-0.15240 0.5334	0.96936 0.0001		
DISLO	0.72512 0.0004	0.70275 0.0008	0.74307 0.0003	0.01727 0.9440	0.82687 0.0001	0.61882 0.0047	0.34516 0.1478	-0.02767 0.9105	0.65043 0.0026	0.73837 0.0005	0.88556 0.0003	-0.14511 0.5534	0.88872 0.0001		
MACH	0.33729 0.1579	0.16211 0.5073	0.15438 0.5280	0.82387 0.0001	0.17251 0.4800	0.14875 0.5433	0.34802 0.1443	-0.03070 0.9007	0.16482 0.5001	0.06959 0.7838	-0.13522 0.6918	0.26467 0.2735	0.21242 0.3826		
TMACH	0.09015 0.7136	-0.01080 0.9650	-0.02204 0.9286	0.31234 0.1930	-0.08031 0.7438	-0.00042 0.9986	0.06069 0.8051	-0.00522 0.9831	-0.16209 0.5074	-0.07047 0.7811	-0.60426 0.0490	1.00000 0.0000	-0.14994 0.5401		
MACH6	0.30776 0.1999	0.12237 0.6177	0.10794 0.6600	0.87711 0.0001	0.12451 0.6115	0.12152 0.6202	0.31324 0.1916	-0.11339 0.6439	0.12653 0.6057	0.07347 0.7720	-0.16689 0.6238	0.29416 0.2215	0.17339 0.4778		
F1	0.84583 0.0001	0.83365 0.0001	0.84829 0.0001	0.15912 0.5153	0.85673 0.0001	0.81099 0.0001	0.31043 0.1958	-0.40076 0.0891	0.69194 0.0010	0.80155 0.0001	0.57894 0.0620	0.20866 0.3913	0.90740 0.0001		
F2	0.88044 0.0001	0.84928 0.0001	0.84584 0.0001	0.19667 0.4197	0.86918 0.0001	0.83452 0.0001	0.23907 0.3243	-0.47037 0.0421	0.81245 0.0001	0.90649 0.0001	0.86743 0.0005	-0.12001 0.6246	0.97752 0.0001		
F3	0.86046 0.0001	0.86558 0.0001	0.87861 0.0001	0.08299 0.7355	0.90433 0.0001	0.84173 0.0001	0.23710 0.3284	-0.42490 0.0698	0.81943 0.0001	0.92477 0.0001	0.92252 0.0001	-0.16626 0.4963	0.99640 0.0001		
F4	0.86563 0.0001	0.81501 0.0001	0.83862 0.0001	0.26621 0.2706	0.84945 0.0001	0.78287 0.0001	0.33174 0.1653	-0.26119 0.2801	0.77375 0.0001	0.86568 0.0001	0.86066 0.0007	-0.04759 0.8466	0.94254 0.0001		
F5	0.77495 0.0001	0.67820 0.0014	0.67488 0.0015	0.23625 0.3302	0.63452 0.0035	0.67330 0.0016	0.18344 0.4522	-0.32912 0.1689	0.56792 0.0112	0.69888 0.0013	0.19134 0.5730	0.58925 0.0079	0.68491 0.0012		

TABLE 2.1 (Concluded)

CORRELATION COEFFICIENTS / PROB > IRI UNDER H0:RHO=0 /														NUMBER OF OBSERVATIONS				
	LOGA	LOGTB	LOGS	LOGMD	DISLO	MACH	TMACH	MACH6	F1	F2	F3	F4	F5					
LOGA	1.00000 0.00000 19	0.94396 0.00000 19	0.95947 0.00000 19	0.93423 0.00000 19	0.90700 0.00000 19	0.18976 0.4365 19	-0.14272 0.5600 19	0.13980 0.5681 19	0.89238 0.0001 19	0.92411 0.0001 19	0.97157 0.0001 19	0.95266 0.0001 19	0.67297 0.0016 19					
LOGTB	0.94396 0.00000 19	1.00000 0.00000 19	0.93472 0.00000 19	0.92546 0.00000 19	0.86439 0.00000 19	0.16785 0.4922 19	-0.23415 0.3346 19	0.12995 0.5960 19	0.78322 0.0001 19	0.93881 0.0001 19	0.95335 0.0001 19	0.89429 0.0001 19	0.64393 0.0029 19					
LOGS	0.95947 0.00000 19	0.93472 0.00000 19	1.00000 0.00000 19	0.91059 0.00000 19	0.93432 0.00000 19	0.30681 0.2014 19	-0.15392 0.5293 19	0.26795 0.2674 19	0.88978 0.0001 19	0.96577 0.0001 19	0.98347 0.0001 19	0.95200 0.0001 19	0.65489 0.0023 19					
LOGMD	0.93423 0.00000 19	0.92546 0.00000 19	0.91059 0.00000 19	1.00000 0.00000 19	0.75781 0.0002 19	0.14802 0.5453 19	-0.15240 0.5334 19	0.12578 0.6079 19	0.88673 0.0001 19	0.95682 0.0001 19	0.96942 0.0001 19	0.87958 0.0001 19	0.65673 0.0023 19					
DISLO	0.90700 0.00000 19	0.86439 0.00000 19	0.93432 0.00000 19	0.75781 0.0002 19	1.00000 0.00000 19	0.25297 0.2961 19	-0.14511 0.5534 19	0.17839 0.4650 19	0.79363 0.0001 19	0.84466 0.0001 19	0.88193 0.0001 19	0.89103 0.0001 19	0.60045 0.0066 19					
MACH	0.18976 0.4365 19	0.16785 0.4922 19	0.30681 0.2014 19	0.14802 0.5453 19	0.25297 0.2961 19	1.00000 0.00000 19	0.26467 0.2735 19	0.97559 0.0001 19	0.29190 0.2253 19	0.32545 0.1739 19	0.21891 0.3679 19	0.47894 0.0380 19	0.30725 0.2007 19					
TMACH	-0.14272 0.5600 19	-0.23415 0.3346 19	-0.15392 0.5293 19	-0.15240 0.5334 19	-0.14511 0.5534 19	0.26467 0.2735 19	1.00000 0.00000 19	0.29416 0.2215 19	0.20866 0.3913 19	-0.12001 0.6246 19	-0.16626 0.4963 19	-0.04759 0.8466 19	0.56925 0.0079 19					
MACH6	0.13980 0.5681 19	0.12995 0.5960 19	0.26795 0.2674 19	0.12578 0.6079 19	0.17839 0.4650 19	0.97559 0.0001 19	0.29416 0.2215 19	1.00000 0.00000 19	0.26339 0.2759 19	0.29317 0.2199 19	0.18678 0.4439 19	0.42367 0.0707 19	0.29942 0.2130 19					
F1	0.89238 0.0001 19	0.78322 0.0001 19	0.88978 0.0001 19	0.88673 0.0001 19	0.79363 0.0001 19	0.29190 0.2253 19	0.20866 0.3913 19	0.26339 0.2759 19	1.00000 0.00000 19	0.89562 0.0001 19	0.90565 0.0001 19	0.82699 0.0001 19	0.82338 0.0001 19					
F2	0.92411 0.0001 19	0.93881 0.0001 19	0.96577 0.0001 19	0.95682 0.0001 19	0.84466 0.0001 19	0.32545 0.1739 19	-0.12001 0.6246 19	0.29517 0.2199 19	0.89562 0.0001 19	1.00000 0.00000 19	0.98040 0.0001 19	0.92584 0.0001 19	0.68536 0.0012 19					
F3	0.97157 0.0001 19	0.95335 0.0001 19	0.98347 0.0001 19	0.96942 0.0001 19	0.88193 0.0001 19	0.21891 0.3679 19	-0.16626 0.4963 19	0.18678 0.4439 19	0.90565 0.0000 19	1.00000 0.00000 19	0.98040 0.0001 19	0.93528 0.0001 19	0.66535 0.0019 19					
F4	0.95266 0.0001 19	0.89429 0.0001 19	0.93528 0.0001 19	0.87958 0.0001 19	0.89103 0.0001 19	0.47894 0.0380 19	-0.04759 0.8466 19	0.56925 0.0079 19	0.88699 0.0001 19	0.92584 0.0001 19	0.93528 0.0001 19	1.00000 0.0000 19	0.69477 0.0010 19					
F5	0.67297 0.0016 19	0.64393 0.0029 19	0.65489 0.0023 19	0.65673 0.0023 19	0.60045 0.0066 19	0.30725 0.2007 19	0.56925 0.0079 19	0.29942 0.2130 19	0.82338 0.0001 19	0.68536 0.0012 19	0.66535 0.0019 19	0.69477 0.0010 19	1.00000 0.0000 19					

- LOGTB: Log Tail Blade Area
- LOGS: Log Shaft Horsepower
- LOGMD: Log Main Rotor Disc Area
- DISLO: Main Rotor Disc Loading

Other quantities which are linearly independent of these parameters can be expected to play the roles of second and third variable in the multiple regression analyses. However, it is possible to see more than one of these variables appear in a multiple regression if the two variables are largely independent of each other.

## 2.2 Summary of Best Correlates

The table provided below identifies those single parameters which best correlate with EPNL for the various operational modes.

<u>Level Flyover</u>		<u>Takeoff</u>		<u>Approach</u>	
<u>Parameter</u>	<u>R<sup>2</sup></u>	<u>Parameter</u>	<u>R<sup>2</sup></u>	<u>Parameter</u>	<u>R<sup>2</sup></u>
				LOG TB	.889
F2	.774	LOG TB	.910	LOG W	.867
LOG W	.769	LOG W	.891	LOG MD	.863
LOG MD	.764	F3	.852	F3	.856
F4	.765	LOG A	.850	LOG A	.828
F3	.739	LOG S	.826	F2	.821
LOG TB	.727	LOG MD	.826	M DISC	.778
LOG A	.727	T BLADE	.819	LOG S	.778
				AREA	.774
				T BLADE	.774

The most apparent trend one observes is the much higher correlation for takeoff and approach EPNL values as compared with level flyover EPNL. One plausible reason for this is that the level flyovers are conducted at a higher speed than the approaches or takeoffs. It is reasonable to assume that forces generating noise in higher speed operation are subject to more variant and anomolous aerodynamic influences.

Another observation is the decline of LOG TB to "fifth place" for the higher speed level flyover. This can be attributed to the diminished tail rotor counter-torque load as the speed increased. This occurs as unbalanced airframe slip stream forces tend to counter the main rotor torque.

### 2.3 Plots of Best Single Variable Regression

The three best single variable regression models are plotted in Figures 2.3.1 through 2.3.3 for takeoffs, in Figures 2.3.4 through 2.3.6 for approach, and in Figures 2.3.7 through 2.3.9 for level flyover. Each plot includes identification of helicopter type and the line of regression.

### 3.0 STEPWISE REGRESSION ANALYSES

This analysis investigates the improvement in prediction accuracy associated with adding a second and third variable to the single parameter regression equation. For the correlation coefficient to

improve the added variables (or steps) must be largely independent of previous step(s) but still related to the level of noise. Three-step models are presented below for takeoff, approach, and level flyover.

### Approach

#### 1 Step

$$\text{EPNL} = 10.2 \log (\text{TB}) + 85.9$$

$$R^2 = .89$$

#### 2 Step

$$\text{EPNL} = 7.1 \log (\text{TB}) + .0007 (\text{M DISC}) + 86.9$$

$$R^2 = .93$$

#### 3 Step

$$\text{EPNL} = 7.9 \log (\text{TB}) + .005 (\text{T SPEED}) + .0006 (\text{M DISC})$$

$$R^2 = .94$$

### Level Flyover

#### 1 Step

$$\text{EPNL} = 9.4 (\text{F2}) + 37.4$$

$$R^2 = .76$$

#### 2 Step

$$\text{EPNL} = 10.4 \log (\text{MD}) + 3.6 (\text{F5}) + 59.5$$

$$R^2 = .84$$

#### 3 Step

$$\text{EPNL} = 10.7 \log (\text{MD}) + 17 (\text{MACH}) + 3 (\text{F5}) + 44.9$$

$$R^2 = .85$$

### Takeoff

#### 1 Step

$$\text{EPNL} = 9.6 \log (\text{TB}) + 83.87$$

$$R^2 = .91$$

#### 2 Step

$$\text{EPNL} = 9.55 \log (\text{TB}) - 7.3 (\text{MACH})^6 + 86.1$$

$$R^2 = .93$$

#### 3 Step

$$\text{EPNL} = 6.3 \log (\text{TB}) + 5.5 \log (\text{MD}) - 9.6 (\text{MACH})^6 + 71.97$$

$$R^2 = .95$$

The following table provides a summary of parameters and correlation coefficients associated with each step of the multiple regression analyses for the various operational modes:

<u>Takeoff</u>		
<u>Step</u>	<u>R<sup>2</sup></u>	<u>Parameter(s)</u>
1	.91	Log TB
2	.93	Log TB, (MACH) <sup>6</sup>
3	.95	Log TB, (MACH) <sup>6</sup> , Log (MD)

<u>Approach</u>		
<u>Step</u>	<u>R<sup>2</sup></u>	<u>Parameter(s)</u>
1	.89	Log TB
2	.93	Log TB, M DISC
3	.94	Log TB, M DISC, T SPEED

<u>Level Flyover</u>		
Step	R <sup>2</sup>	<u>Parameter(s)</u>
1	.76	F2
2	.84	F5, Log MD
3	.85	Log MD, MACH, F5

#### 4.0 DISCUSSION

The empirical noise prediction techniques presented above provide a means to estimate single event cumulative noise exposure for helicopters whose technology and operational characteristics are similar to the helicopters used in this analysis. The obvious advantage of the strictly empirical approach (using the single event cumulative energy metric EPNL) is the averaging out of source directionality, ground interference effects, anomolous air in-flow characteristics, and speed effects. All of these considerations pose significant difficulties in the theoretical approach. On the other hand, the theoretical approach can be more successful in predicting the change in noise level associated with a design change for a specific helicopter.

The parameters appearing (or not appearing) as significant correlates to noise in this study raise some interesting questions. For example, the strength of LOG TB was not anticipated nor is "Tail Blade Area" a particularly good acoustical design parameter. On the other hand, the main rotor advancing tip mach number to the sixth power (considered an important correlate to noise) appears as a weak correlate by itself, although it does strengthen the takeoff correlation in the multiple regression analysis. Interpretation of these results and discussion of application in predicting noise of new design helicopters or reducing levels of existent machines is left for subsequent study.

# TAKEOFF REGRESSION OF EPNL VERSUS LOGTB EMPIRICAL NOISE PREDICTION ANALYSIS

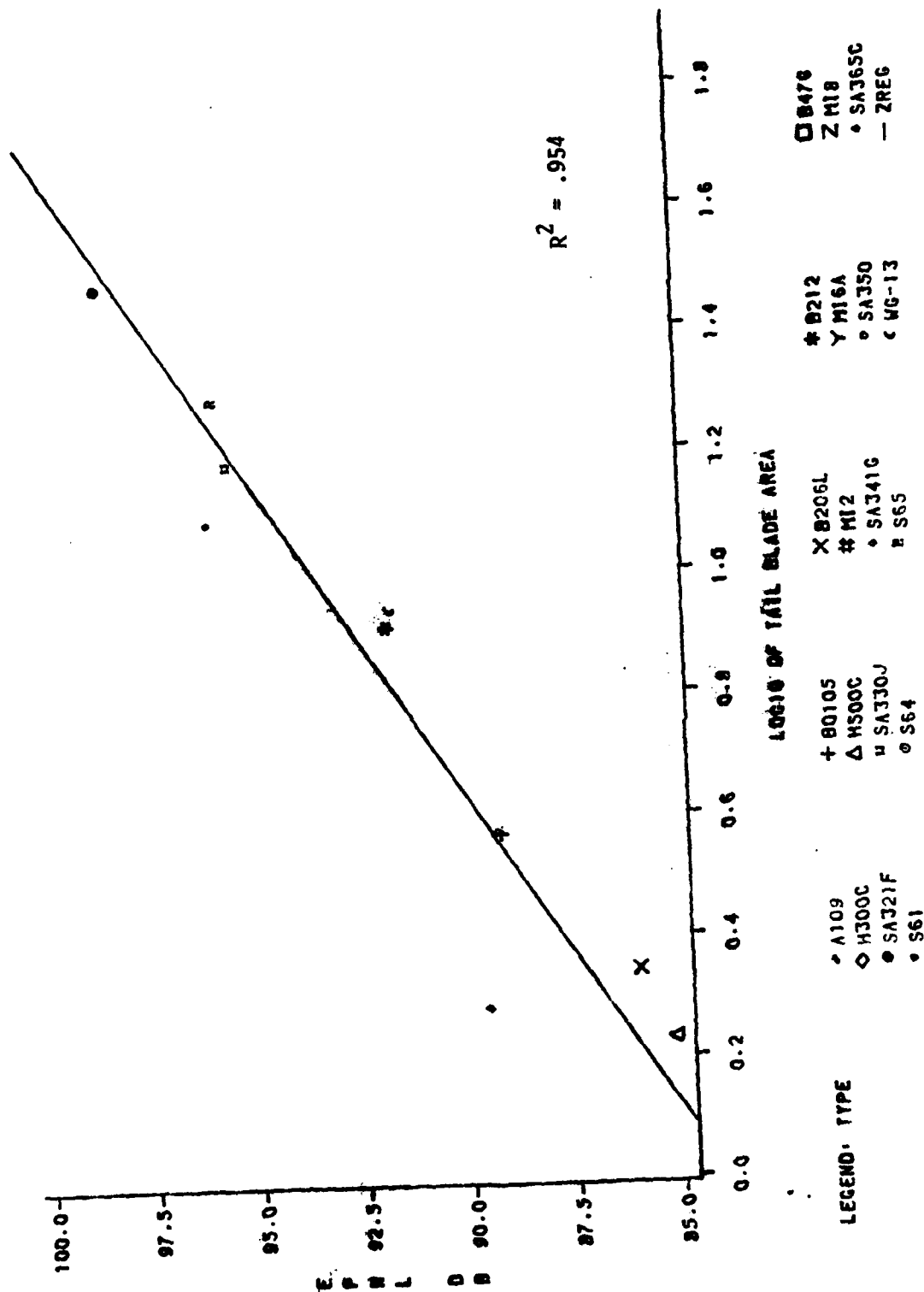


FIGURE 2.3.1



# TAKEOFF REGRESSION OF EPNL VERSUS LOGW EMPIRICAL NOISE PREDICTION ANALYSIS

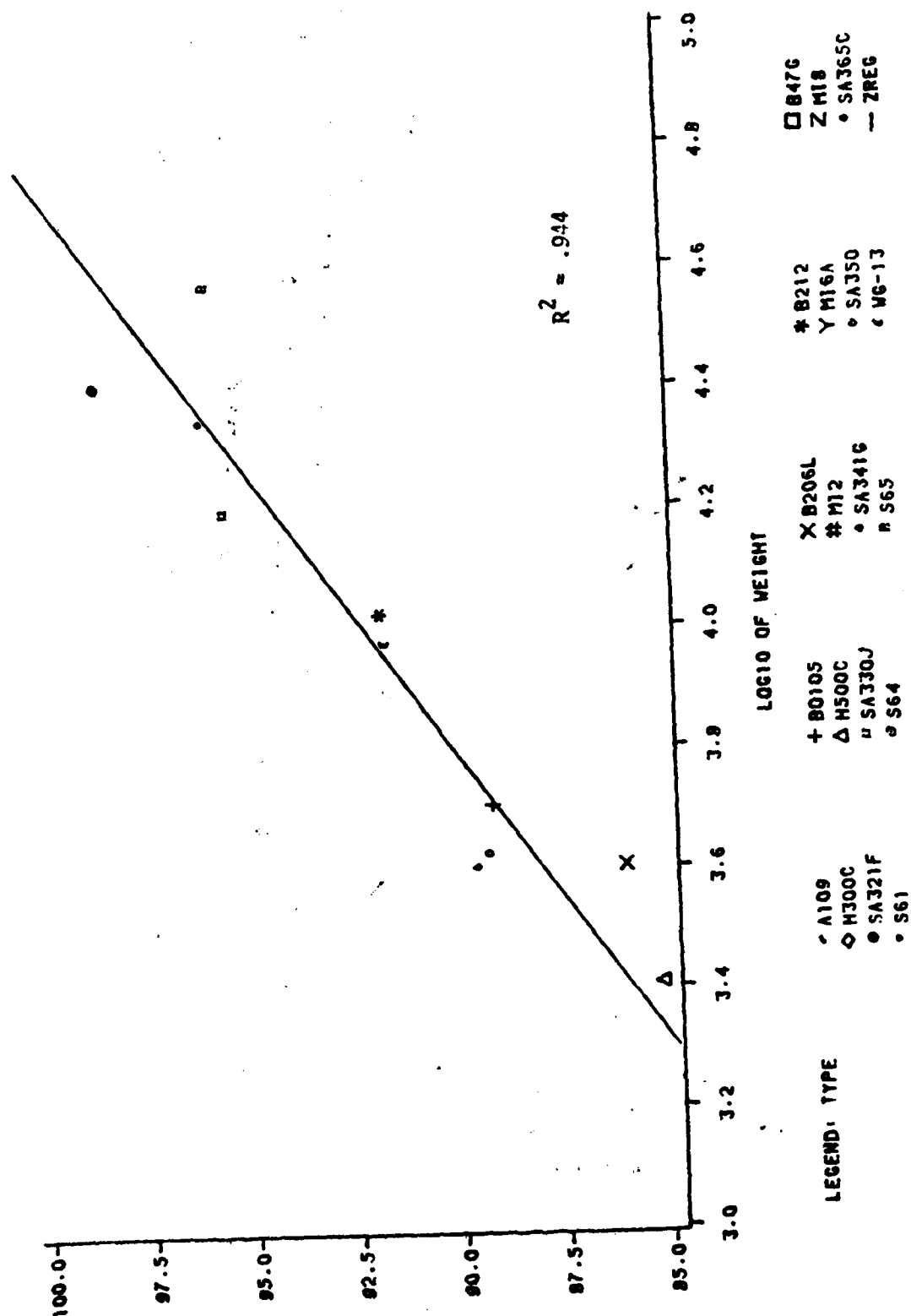
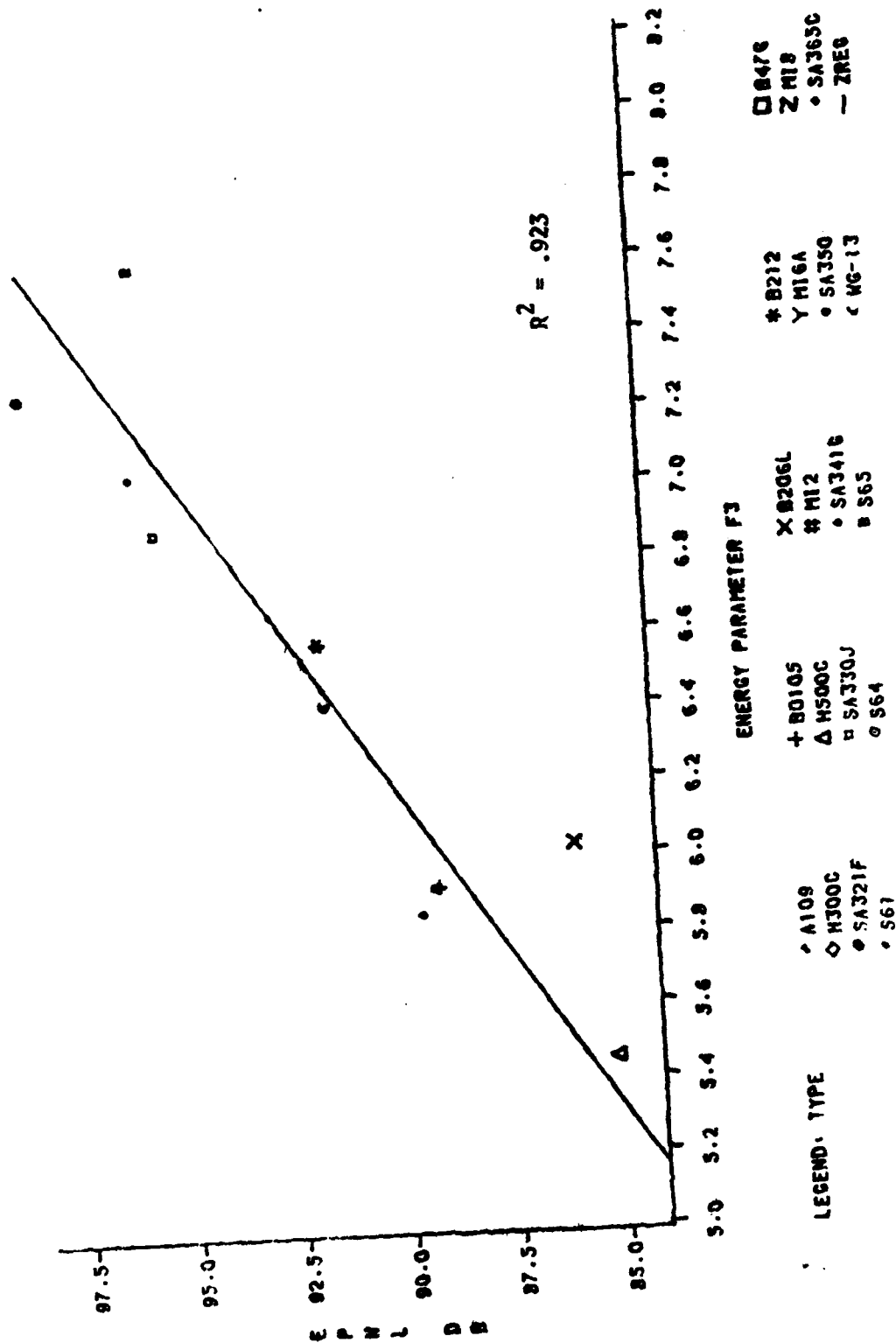


FIGURE 2.3.2

# TAKEOFF REGRESSION OF EPNL VERSUS F3 EMPIRICAL NOISE PREDICTION ANALYSIS



F3=LOG10 (SHP-MAIN DISC AREA/MACH)

FIGURE 2.3.3

# LEVEL FLYOVER REGRESSION OF EPNL VERSUS F2 EMPIRICAL NOISE PREDICTION ANALYSIS

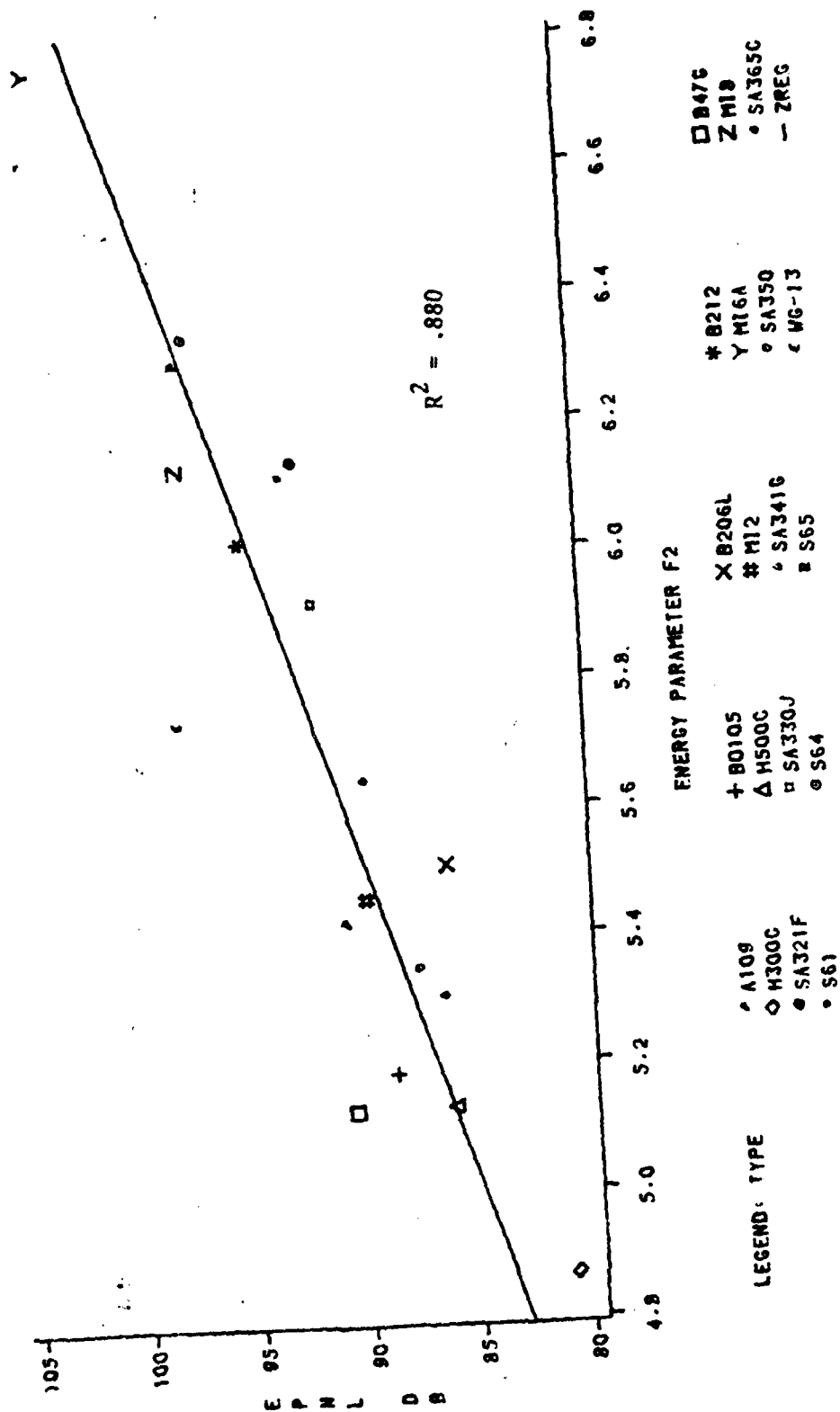


FIGURE 2.3.4

F2=L0010 (MACH\*WEIGHT)\*\*2/AREA)

# LEVEL FLYOVER REGRESSION OF EPNL VERSUS LOGW EMPIRICAL NOISE PREDICTION ANALYSIS

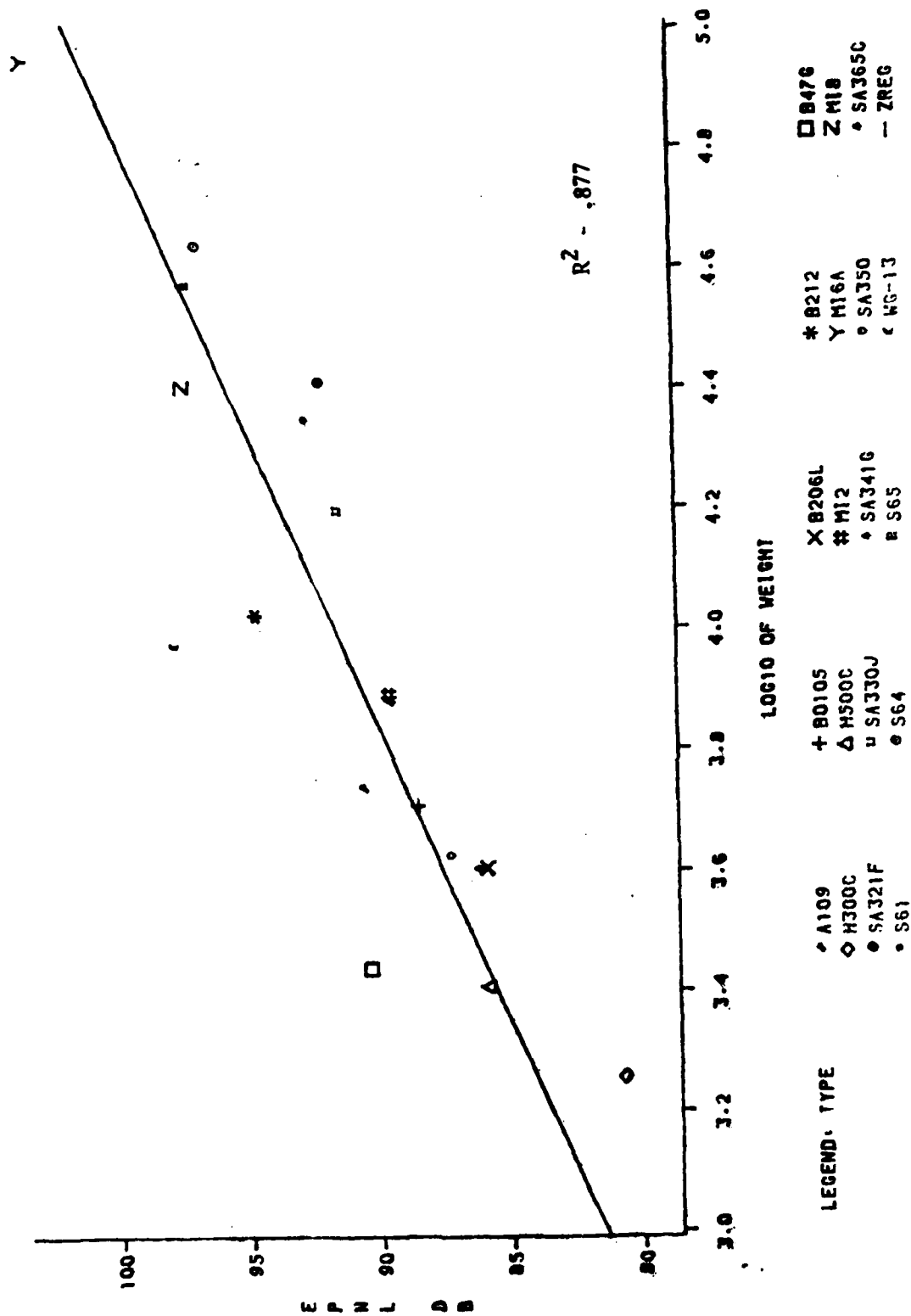


FIGURE 2.3.5

# LEVEL FLYOVER REGRESSION OF EPNL VERSUS LOGMD EMPIRICAL NOISE PREDICTION ANALYSIS

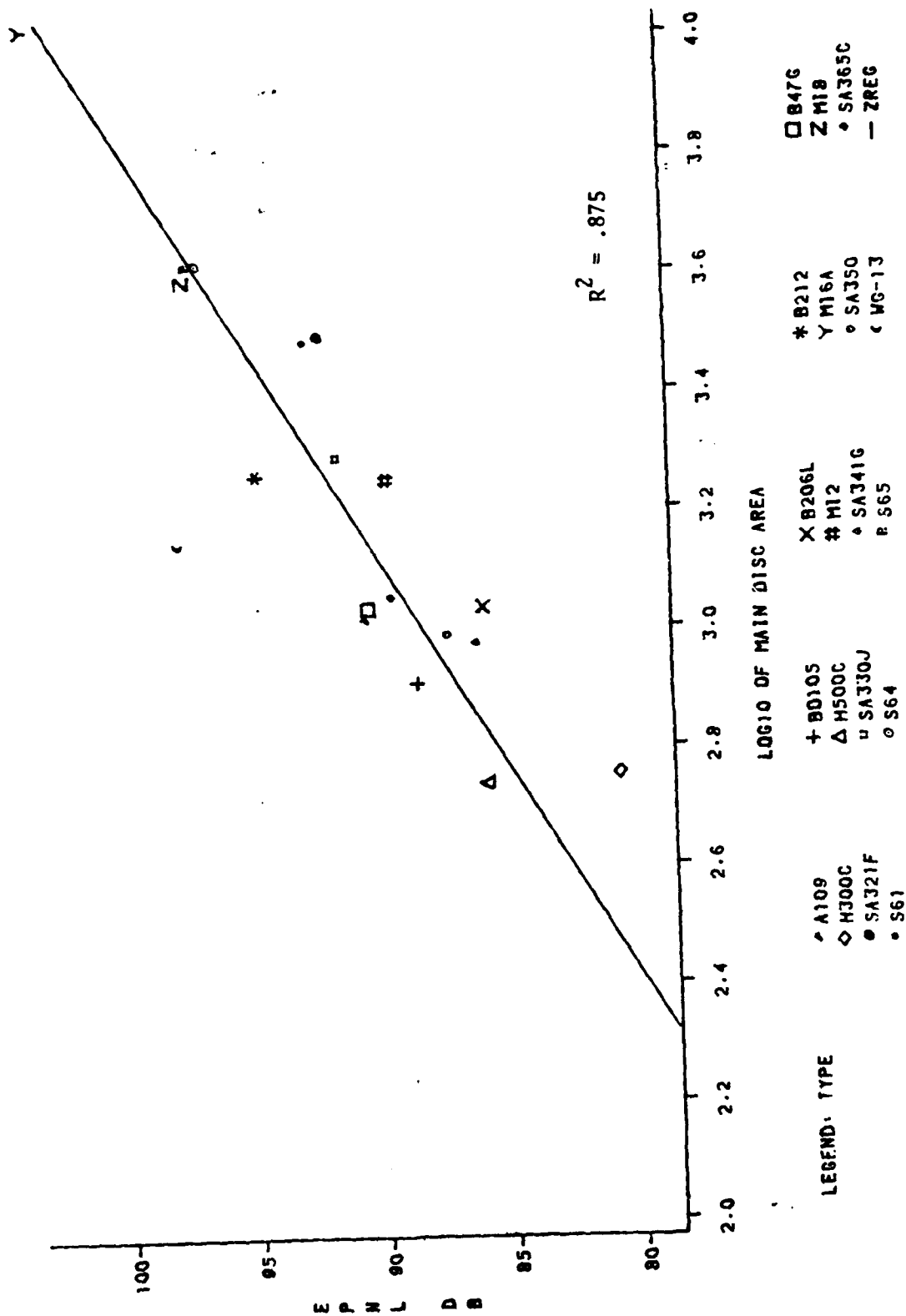


FIGURE 2.3.6

# APPROACH, REGRESSION OF EPNL VERSUS LOGTB EMPIRICAL NOISE PREDICTION ANALYSIS

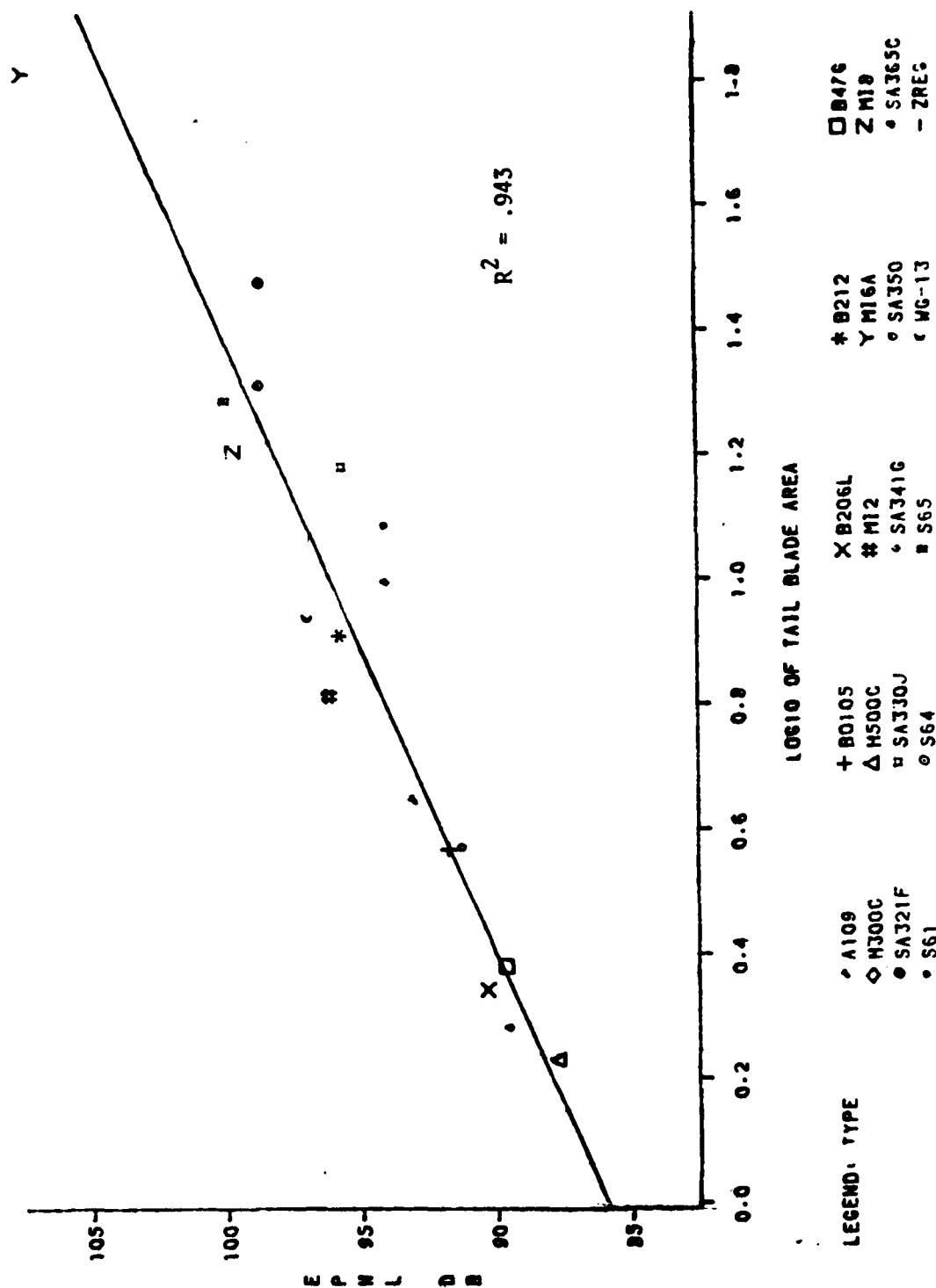


FIGURE 2.3.7

# APPROACH, REGRESSION OF EPNL VERSUS LOGW EMPIRICAL NOISE PREDICTION ANALYSIS

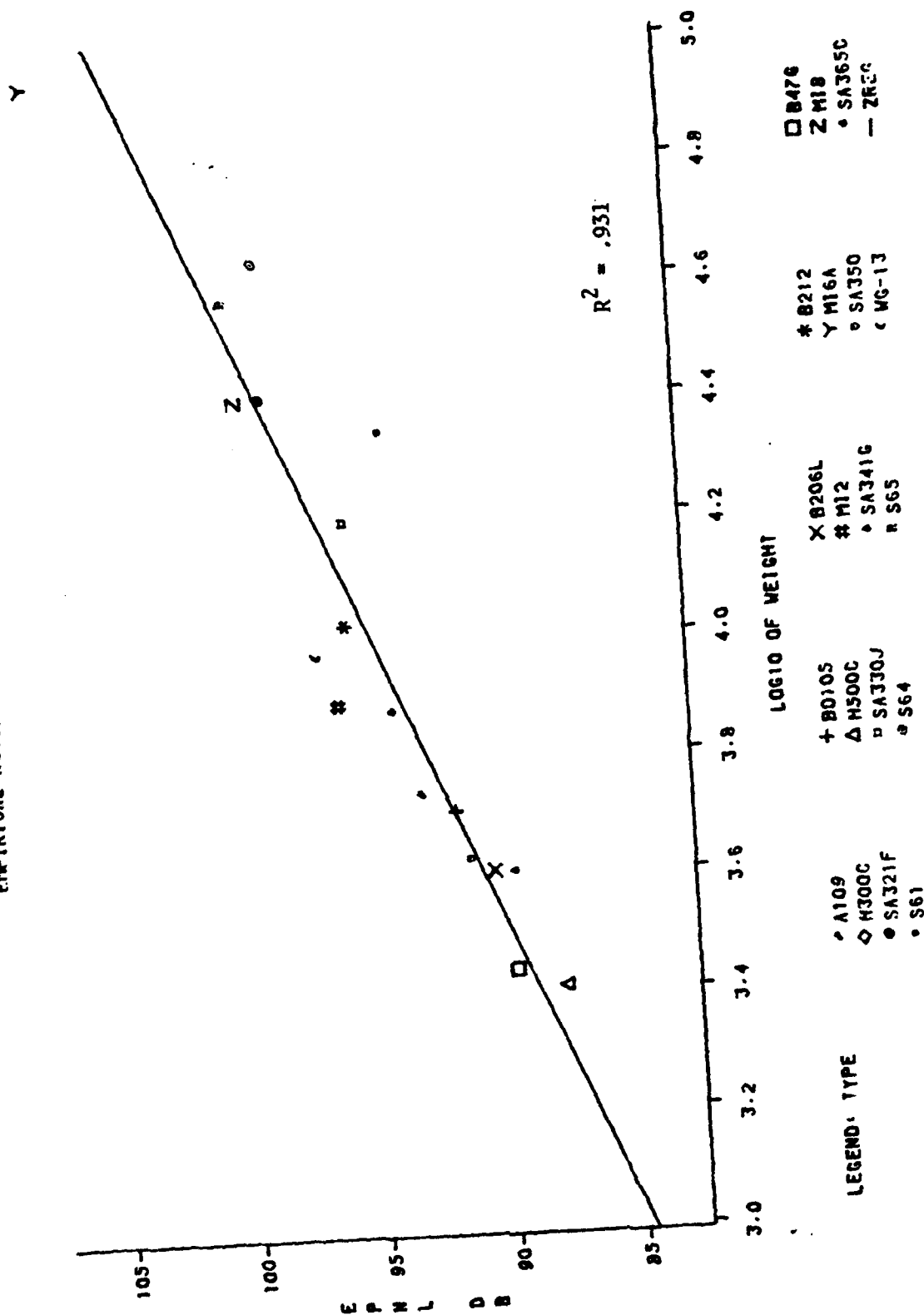


FIGURE 2.3.8

# APPROACH REGRESSION OF EPNL VERSUS LOGMD EMPIRICAL NOISE PREDICTION ANALYSIS

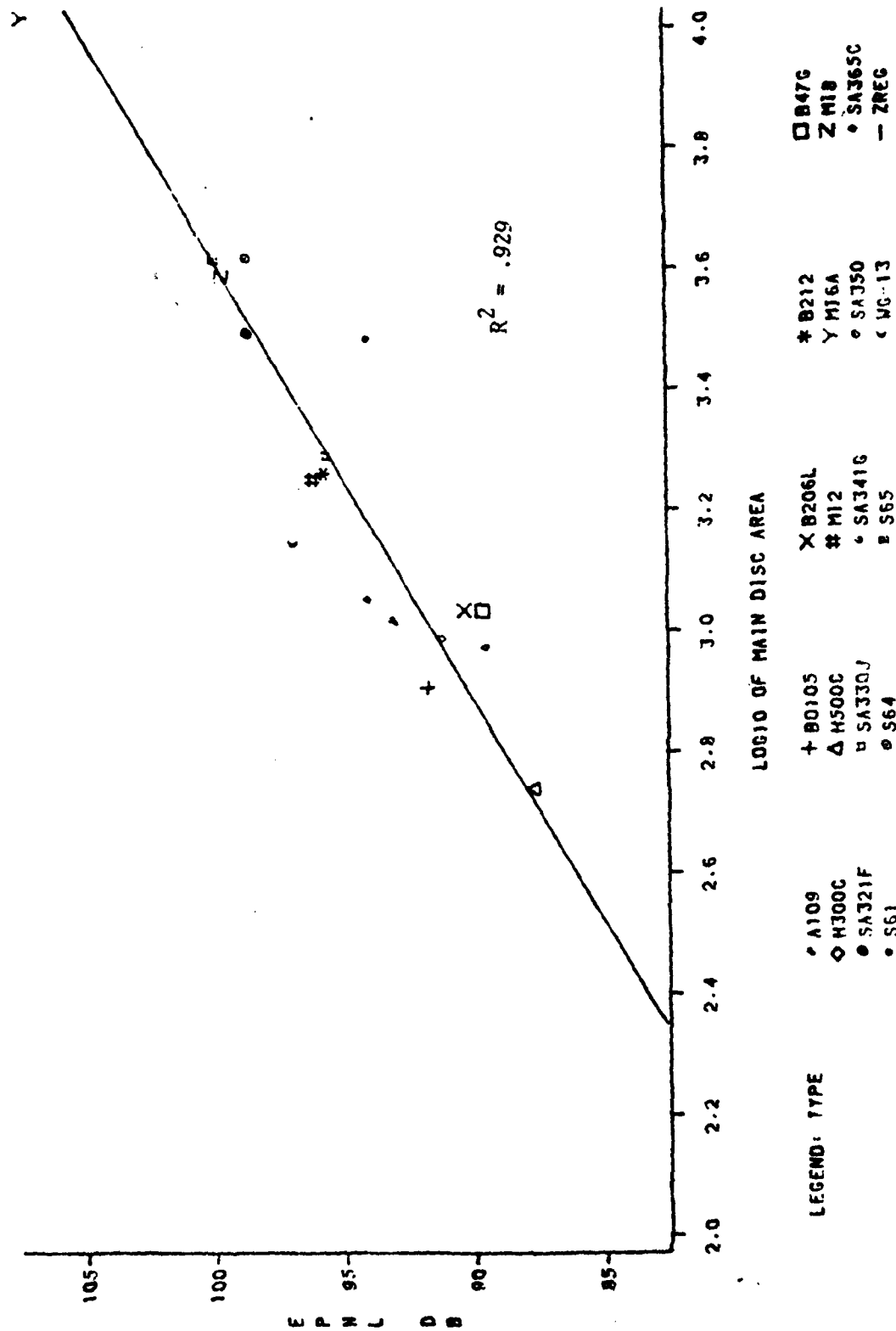


FIGURE 2.3.9



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